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for a Two-Stage Nonrecoverable Space Rocket,
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by N. N. Nikitin, 13 pp.
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FTD-S-~~386~~ 61-69

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apr 69

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Phenomena in Testing Centrifugal Pumps on Gases,
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Instability of the Zone of Turbulent Burning in a Rocket Engine Relative to a Change in ~~its~~ Its Length, by S. K. Aslanov.

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N. I. Sapozhnikov.

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NASA TT F-12,136

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Criterion of Passenger Aircraft Efficiency,
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May 69

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Which engine has maximum theoretical range for a
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AVIA TT F-12,732

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Determining Aircraft and Engine Parameters,
Assuring Maximim Theroetical Range at
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21-11-~~2~~59

BT69-1-2154T

Drag estimation in laval nozzles

B.S. Vinogradov, M.K. Makutova

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6/2/69*

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O. N. Yemin. 12 pp.

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NASA TT F-12,980

NTC 72-14337-202

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nov 70

REQ TR CHEKC

19-03-73

T373

INFLUENCE OF COOLING ON THE WORKING FLUID
PARAMETERS IN AN AIR-COOLED TURBINE STAGE
KOKAY, V I; KUMIROV, B A

UR: IZVESTIYA VYSSHIKH UCHEBNYKH ZAVEDENIY,
AVIATSIONNAYA TEKHNKA, 1969, VOL 12, NO
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Process in air

H573-23-0365

*Neg
28 MAR 73*

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N. V. Pervyashin. 9 pp.

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NASA TT F-12,983

NTC 72-14345-20M

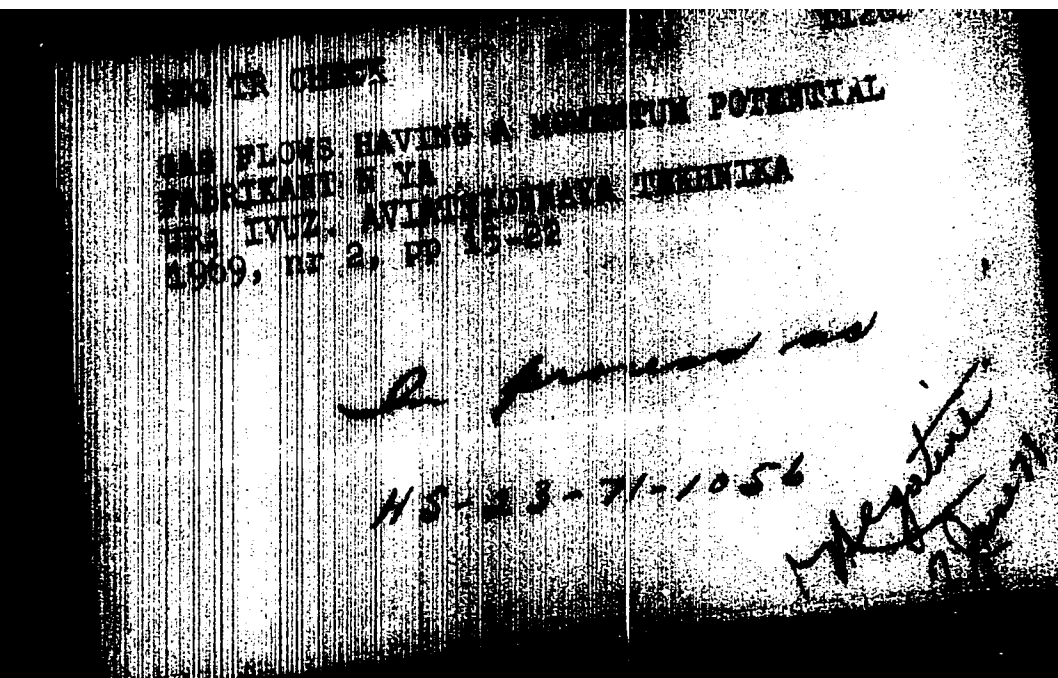
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*FTD IT Special 136-70

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FTD HT Special 150-70

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with a Supercritical Boiling Regime, by

Yu. P. Yanovlev.

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*NASA TT F 12,974

FTD AT 7-10-69

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apr 70

70-13585-20K

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pp 143-148.
FTD HT Special 151-70

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June 70

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RUSSIAN, per, Izv VUZ Avlats Tekh, No 2, 1969,
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June 70

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of Design Outlays for a Variable Geometry Wing,
by V. P. Goglin.

RUSSIAN, per, Izvestiya VUZ Aviatsionnaya Tekhnika,
Vol 12, No 2, 1969, pp 169-171.

*FTD HT Special 133-70

May 70

Optimum Control of Boundary Layer, by O. G. Divakov,
T. K. Syrase'dinev.

RUSSIAN, per, Izvestiya VUZ Aviatstomnaya Tekhnika,
No 3, 1969, pp 5-13.

*FTD HT Special 135-70

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To the Problem of Laminar Boundary Layer Steadiness of Noncompressible Fluid in Account of Transverse Speed, by M. V. Tzvetkova.

RUSSIAN, per, Izv VUZ Aviats Tekh, No 3, 1969, pp 33-40.

FTD HT Special 153-70

Sci-Phys

June 70

Karasev, O.V.

Determining the Angle of Downstream
Flow from a Two-Dimensional Turbine Blade
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IZVESTIYA VYSSIKH UCHEBNYKH ZAVEDENII,
AVIATSIONNAYA TEKHNKA, N. 2, 1968,

P. 66-71

NTC-72-14627-20D

Feb 73

REQ TR CHECK

20-03-73

T375

CALCULATION OF THE TEMPERATURE FIELD OF
COOLED TURBINE BLADES FOR VARIABLE BOUNDARY
CONDITIONS AS A FUNCTION OF THE THERMOPHYSICA
PROPERTIES OF THE MATERIAL AND THE
TEMPERATURE

LOKAY, V.I

UR: IZVESTIYA VYSSHIKH UCHEBNYKH ZAVEDENIY,
AVIATSIONNAYA TEKHNIKA, 1969, VOL 12, NO 3,
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*Process in a**HS72-23-0366**neg
28 MAR 73*

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Scheme of Heat Transfer with Supercritical Regime
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Turbulent Boundary Layer Parameters, by
G. V. Filippov, V. G. Shakhov.
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NTC-70-13156-20K

CIA 70-13156-20K

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*Determination of Stress and Strain Components in the
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Theoretical Performance of a Turbojet Engine Nozzle
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I. N. Dyatlov.
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No 4, 1969, pp 153-157.
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Vol 12, No 4, 1969, pp 158-161
FTD HT Special 327-70

sci
nov 70

REQ TR CHECK

3-11-25

TW84

INVESTIGATION OF COMBUSTION IN THE
AFTERBURNER CHAMBER OF A TURBOPROP
ENGINE WITH AIR FUEL FLAME STABILIZERS

BULAVKIN A A

UR: IVUZ. AVIATIONNAYA TEKHNIKA, 1969,
nr 4, pp 59-67

In process as

MS-24-71-1957

Negative
9 Nov 71

Prospects for Cooling with Diathermic Heat
Exchange Systems, by Yu. P. Yakovlev.
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No 1, 1970, pp 28-42
FTD HT Special 321-70

nov 70

101 TR CHECK

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THE BUREAU OF INVESTIGATION AND THE FBI IS
INTERESTED IN THE FACTS OF
THESE CASES AND IN THE NAMES OF
THE

GARBUNOV G M

VR: IVUZ, AVIATSIONNAYA TEKHNIKA, 1970,
nr 1, pp 88-96

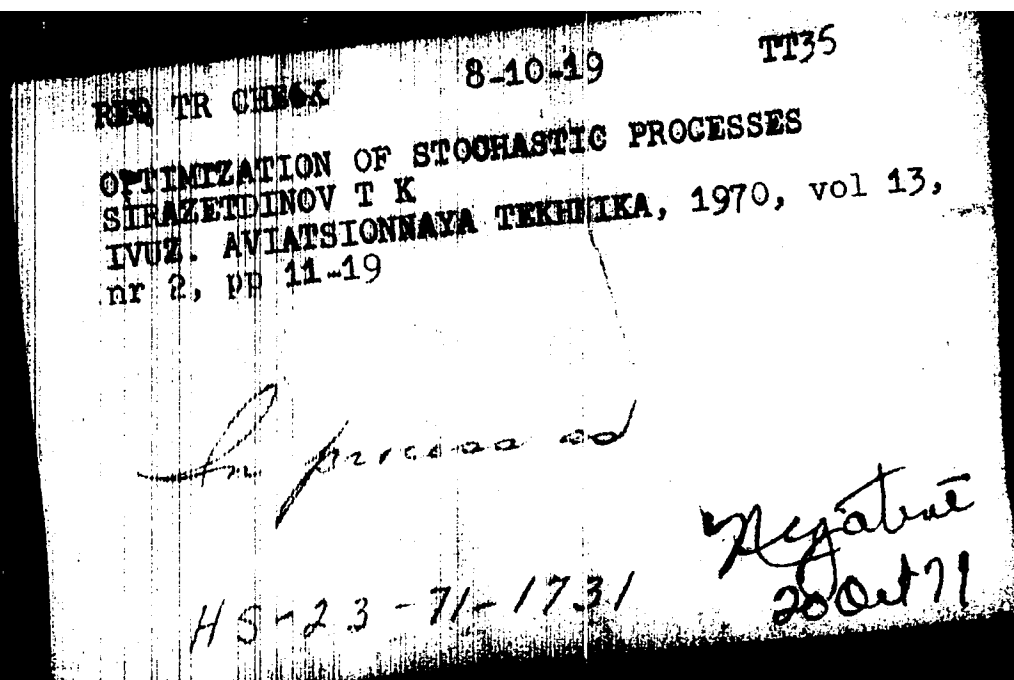
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10-23-71-1055

Flame Propagation Velocity in Turbulent Flow
Under Two-Stage Combustion, by V. A. Shchukin,
A. V. Mosin.

RUSSIAN, per, Izvestiya Vysshikh Uchebnykh
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1970, pp 138-146.

FTD HT Special 304-70

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Unsteady Heat Conductivity of a Cylindrical Wall for
A Variable Heat Transf....., by Yu. V. Vidin.
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No 2, 1970, pp 52-57.
*RSTC-ET-23-1047-72

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The Stationary Temperature Fields of the Wall of a
Pipe and the Heat Carrier for Small Values of the Bi
Criterion, by S. M. Galitskiy, G. A. Breytser, 12 pp.
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No 2, 1970, pp 98-98.
AM/STC/ST-23-15-72

Determination of the Aerodynamic Characteristics
of an Elastic Swept Wing in Subsonic Flow, by
A. Kh. Karimov. 15 pp.

RUSSIAN, per, Izv. Vysishikh Uchebn. Zavedenii,
Moscow, Vol 15, No 5, 1970, pp 5-15.
REF ID: A13717

47D HD-23-809-71
NTC 72-10690-01A

REQ TH ORDECK

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TG53

FLOWING AT AERIAL WIDE-ANGLE PASSAGE
GORELOV G M; TRYANDY A YR
UR: IVOZ, AVIATSIONNAYA TEKHNIKA,
1970, nr 3, pp 54-62

*In process as**HS-23-71-0806**Negative*
1970

REQ FOR CHECK

26-5-39

FI24

ON A PROBLEM OF DESIGN POINT ON TURBOPAN
PARAMETERS WITH HIGH BYPASS
PIKOVSHIN N V; KUCHINSKAYA T P
UR: IVUZ. AVIATIONNAYA TEKHNIKA
1970, nr 3, pp 63-69

*In process**H.S.-23-71-1054**Negative
June 71*

Avdeev, V. B.

Experimental Study of the Energy Losses of
A Radial Flow in a Rotating Circular Cascade
of Cylindrical Profiles of a Centripetal
Turbine Rotor.

IZVESTIYA VYSSHIKH UCHEBNYKH ZAVEDENII.
AVIATSIONNAYA TEKHNIKA, No. 3., 1970, p.75-85
NTC 72-14628-20D

Feb 73

REQ TR CHECK

20-03-73

TAVV

EXPERIMENTAL INVESTIGATION OF TURBINE STAGE
WITH AN EXHAUST OF COOLING AIR INTO THE
FLOW AREA

LOKAY, V I; KUMIROV, B A

UR: IZVESTIYA VYSSHIKH UCHEBNYKH ZAVEDENIY,
AVIATSIONNAYA TEKHNIKA, 1970, VOL 13, NO 3,
pp. 93-100

*Presented in 1971**4573-23-0047**289
28 MAR 73*

AT SMALL ANGLES OF ATTACK BY ANALYTICAL
METHOD

PRIVARNIKOV O A

UR: IVUZ. AVIATSIONNAYA TEKHNKA
1970, nr 3, pp 131-135

In process

HS-23-71-0805

*Negotiated
19 May 71*

REQ TR CHECK

7-2-6

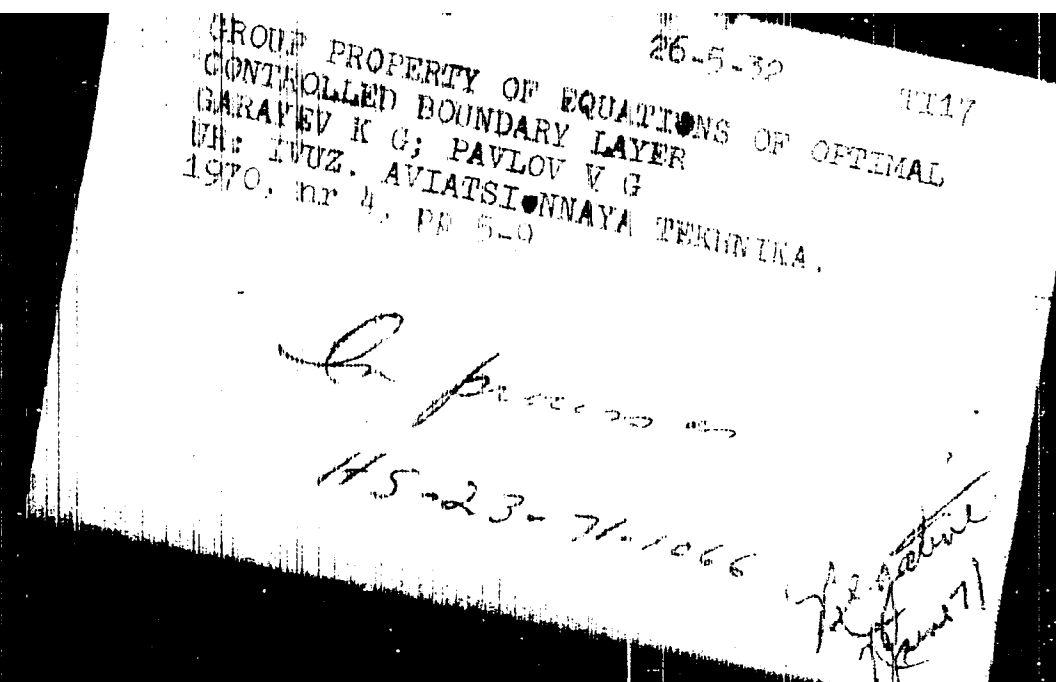
CURRENT LINER IN COMPLEX FLAT SUPERSONIC
FLOW

TSYBIZOV YU I

UR: IVUZ. AVIATIONNAYA TEKHNIKA, 1970,

nr 3, pp 135-140.

*In process**45-23-71.0804**Negative*
19 May 71



REQ TR CHECK

26-5-30

TH15

CALCULATION OF LIFT FOR THE WING WITH A
ROTATING FLAP

PAPEL T S

UR: IVUZ. AVIATIONNAYA TEKHNIKA

1970, nr 4, pp 10-15

*Process as**H/S-23-71-1068**Heigatun
H/S-23-71*

REQ TR CHECK

26-5-33

TI18

TEMPERATURE REGIME OF AIRCRAFT CONSTRUCTION
ELEMENTS UNDER AERODYNAMIC HEATING

SALOMATOV V V

UR: IVUZ. AVIATIONNAYA TEKHNIKA, 1970,
nr. 4, pp 22-30

In process as
145-23-71-1065

Signature
27

PRISNYAKOV V F
UR: IVUZ. AVIATSIONNAYA TEKHNIKA
1970. pr 4, pp 72-79

[Handwritten signature]
HS-23-71-1321

Negative
[Handwritten signature]
26/4/71

REG. TR. CHECK

7-11-72

TAEI

INFLUENCE OF SOME DESIGN PARAMETERS OF GAS
TURBINE ENGINE COMBUSTION CHAMBER ON THE
CIRCUMFERENTIAL NONUNIFORMITY OF THE GAS
SPIRIDONOV YU A

UR: IVUZ. AVIATSIONNAYA TEKHNIKA. 1972.
no 4, pp 30-36

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3 FEB 1972

Expansion of a Turbulent-jet Boundary Layer in a
Slipstream, by V. K. Temnov.

RUSSIAN, per, IVUZ Aviatsionnaya Tekhnika, Vol 13,
No 4, 1970, pp 111-115.

NASA-TT-F-14060

NTC 72-10455-8C D

FTD-72-23-7-1067

Nov 71